

**George W. Woodruff School of Mechanical Engineering**

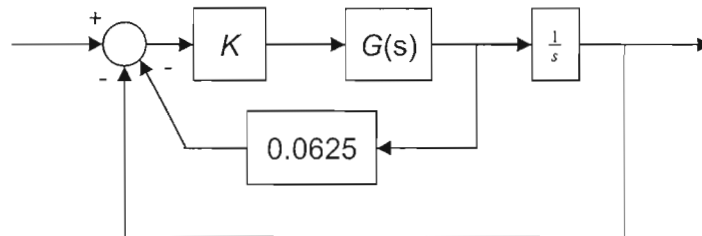
**Spring 2005 System Dynamics and Controls**

**Doctoral Qualifying Examination**

**INSTRUCTIONS**

There are 4 questions attached, please solve 3 of the four questions as completely as possible. State all assumptions, and make sure that you clearly indicate the thought processes that you employed to arrive at your answer. Answer only 3 questions. If you answer 4 questions, only the first 3 will be graded.

## ROOT LOCUS



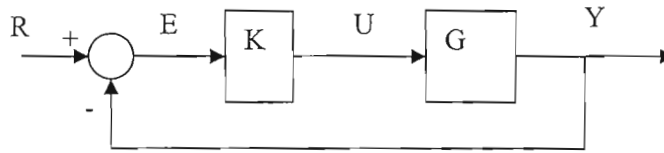
In the velocity-feedback control system shown above,  $K > 0$ , and the plant transfer function is

$$G(s) = \frac{1}{(s^2 + 18s + 162)(s + 9)}$$

- (a) Draw the root locus, with any break-away/break-in points, asymptote intercepts, and imaginary axis crossings accurate to within  $\pm 1$  rad/sec, and angles of departure/arrival and of any asymptotes accurate to within  $\pm 10^\circ$ . Show all calculations.
- (b) For what value of  $K$  will the dominant closed-loop pole be located at  $s = -1$  rad/sec? Estimate the settling time of the closed-loop system step response at this value of the gain.

## FREQUENCY RESPONSE

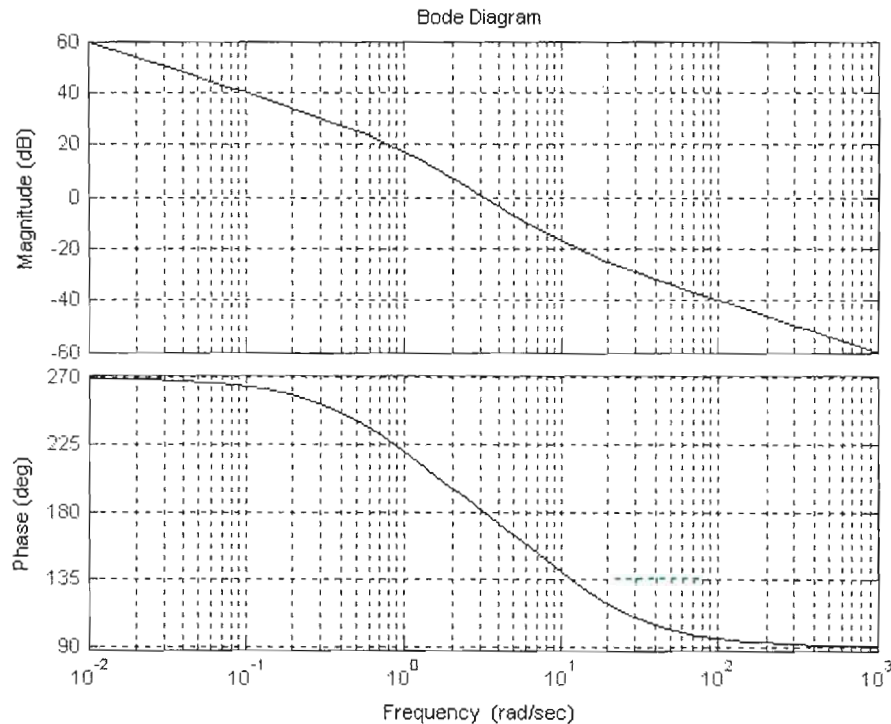
Consider the feedback system shown in the block diagram below.



The Bode diagram of the plant  $G(s)$  is given in the graph below. Note that in the phase plot  $270 = -90$ ,  $180 = -180$ , and  $90 = -270$  degrees.

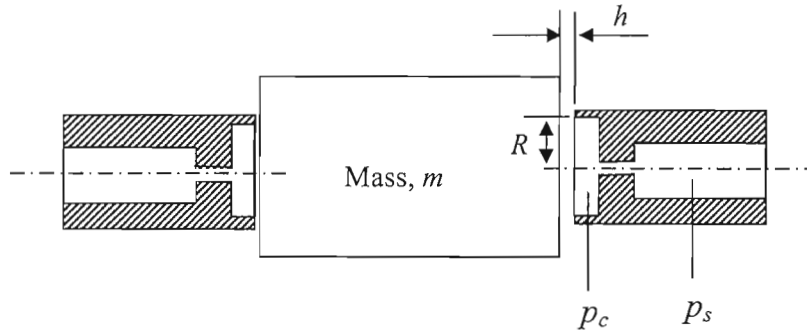
Answer the following questions based on the Bode diagram:

- For what values of the proportional control gain  $K$  is the closed-loop system stable?
- Find the steady-state error in response to the unit ramp reference input  $r(t)=t$  as a function of  $K$ . For what values of  $K$  are your answer valid?
- Repeat part (b) for the sinusoidal reference input  $r(t)=\sin t$
- Estimate the transfer function  $G(s)$ .
- Design a feedback controller (P, PD, or PID) such that the compensated system has no steady-state error to a constant reference input with a phase margin of at least 45 degrees and a cross-over frequency of at least  $\omega_c=10$  rad/sec.



## MODELING

The figure (shown below) illustrates a non-contact bearing system consisting of a pair of pocketed-orifice bearings, each of which is supplied with constant air pressure  $p_s$ .



The pocketed-orifice bearing can be modeled as a dynamic system consists of an air capacitor (with pressure  $p_c$ ) and two orifice resistors (a fixed orifice between  $p_s$  and  $p_c$ , and a variable orifice between  $p_c$  and the ambient). You may assume that the flow through the orifice is proportional to the square root of the pressure difference across it.

- (1) Derive a linearized model to describe the gap motion  $h(t)$ .
- (2) Determine the design criteria such that the non-contact bearing system will function as a self regulator of the gap  $h$ .

## CONTROL DESIGN

A simplified model of a classical naval nuclear reactor is given by the following transfer function:

$$H(s) = \frac{(s+4)}{(s+1)(s-1)}$$

Your mission is to design a controller, P, PI, PD, PID, Lead, Lag, Lead-Lag, etc... to meet the following specifications:

1. Use the simplest controller possible (so if a PI works, using a PID will be incorrect).
2. Use the lowest gains possible.
3. The real part of the poles must be less than -6.
4. The imaginary part of the poles must be zero (*i.e.*, your closed loop system should have poles on the real axis.)

Since this design is going to be used for a nuclear reactor, the Navy has requested that you design the controller using the root locus method (drawing a detailed root locus) and then validate your design by analyzing the closed-loop transfer function.

### Part A

Design the controller

### Part B

In no more than 10 sentences, discuss the relationships between your answers in **Part A** and the closed-loop performance of the system.